

型号认可证数据单
THE VALIDATION DATA SHEET

编号/No: VTC 0231A
版次/Revision: 1
型号/Type:
A380-841 A380-842

批准人/Approved By:



日期/Date: 2014 年 02 月 26 日

本数据单是型号认可证（编号： VTC 0231A ）的组成部分，它规定了符合中国民用航空总局的适航要求所颁发此型号认可证的产品状态和限制。

This data sheet, which is part of **Validation of Type Certificate (No: VTC 0231A)**, prescribes condition and limitation under which the product for which the type certificate was issued meets the airworthiness requirements of the Chinese Civil Aviation Regulation.

型号认可证持有人/Validation of Type Certificate Holder:

Airbus
1 Rond-point Maurice Bellonte
31707 Blagnac
France

有效页清单/List of effective pages :

页数/ Pages	1	2	3	4	5	6	7	8	9	10	11	12	13
版次/Revision	1	1	0	0	0	0	0	1	1	1	1	1	1

中国民用航空局
CIVIL AVIATION ADMINISTRATION OF CHINA

第 1 节 概述（所有型别） / SECTION 1 GENERAL (ALL MODELS)

I. CAAC 认可数据单和型号合格证数据单 / CAAC Validation Data Sheet and Type Certificate Data Sheet

	中国民用航空局认可数据单号、版次和颁发日期 CAAC Validation Data Sheet Number, Revision and issuance date	型号合格证数据单号、版次和颁发日期 Type Certificate Data Sheet Number, Revision and issuance date
当前 Current	0231A, Revision 1, 2014 年 02 月 26 日	EASA TCDS A.110 (Issue 08, 24 September 2013)
历史 History	0231A, Revision 0, 2009 年 12 月 15 日	EASA.110 (Issue 05.0, 1 December 2009)

II. 中文标记标牌/ Chinese markings and placards

所有中文标记标牌应符合如下文件 / All Chinese markings and placards must conform to the following documents:

- A380 External Markings Booklet (ref CC-0800050 dated June 09, 2009)
- A380 Passenger aircraft, passenger and crew placards and illumination signs (ref. L11RQ080315, dated June 9, 2009)

或后续的批准版本 / or later approved revisions.

III. 型号认可证后被中国民航局认可的设计更改:

The following design changes have been validated post Type Validation by CAAC:

改装包 / MOD	描述 / Description
Major Change 62970	CSN01 客舱布局 Install Cabin Layout for CSN01
Major Change 67860	提高隔音隔热材料的阻燃标准—火焰穿透要求 Improved flammability standards for Thermal/acoustic insulation materials - Flame penetration requirements
Major Change 66342	自动驾驶-A380 飞机安装滑跑偏离刹车 (Brake to Vacate) 功能 (第一版) Autoflight-Install Brake to Vacate function on A380 rev 1
Major Change 70408	A380 飞机上定义和安装滑跑偏离刹车 (Brake to Vacate) 功能 (第二步) Define and Install Brake to Vacate function on A380 step 2

第 2 节 A380-800 系列 / SECTION 2 A380-800 SERIES**I. 概述 / General**

1. 飞机 / Aeroplane: A380-800

II. 审定基础 / Certification Basis

1.EASA 合格审定的参考申请日期 / Reference Application Date for EASA Certification:

A380-841/-842: 2001 年 12 月 20 日 / 20 December 2001

2. EASA 合格审定日期 / EASA Certification Date:

A380-841/-842: 2006 年 12 月 12 日 / 12 December 2006

3. EASA 审定基础 / EASA Certification Basis:

参考申请日期有效的下列 EASA/JAA 适航标准:

The following EASA/JAA airworthiness standards effective on the reference date are:

- JAR 1 at change 5 plus orange papers 1/97/1 and 1/99/1
- JAR 25 at change 15
- JAR AWO at change 2 (post TC for autoland)

4. 专用条件 / Special Conditions:

4.1 根据 JAR 21.16(a)(1), 对于适用于 JAR 25 的设计, 产品具有新颖独特设计特点而颁发专用条件

Special Conditions issued because the product has novel or unusual design features relative to the design practices on which the applicable JAR 25 are based (JAR 21.16(a)(1)):

SC B-01	失速和计划运行速度 Stalling and scheduled operating speeds
SC B-02	驾驶舱控制的动作和影响 Motion and effects of cockpit control
SC B-04	静航向、侧向和纵向稳定性以及低能警告 Static directional, lateral and longitudinal stability and low energy awareness
SC B-05	飞行包线保护 Flight envelope protection
SC B-06	正常载荷系数限制系统 Normal load factor limiting system
SC B-10	飞行驾驶舱新特性的人为因素评估 Human factors evaluation of novel features in the flight deck
SC B-15	平稳复飞模式 (TC 证后) Soft Go-Around mode (Post TC)
SC C-01	大飞机结构的适坠性 Crashworthiness of Large Aircraft Structures
SC C-02	离散突风 Discrete gust
SC C-03	多起落架载荷条件

	Loading conditions for multi leg landing gear
SC C-04	起落架车架横向载荷 Undercarriage lateral turning loads
SC C-05	顶起 Jacking by landing gear
SC C-06	动态刹车 Dynamic braking
SC C-11	系统和结构的交联 Interaction of systems and structures
SC C-13	设计机动要求 Design manoeuvre requirements
SC C-15	设计俯冲速度 Vd Design dive speed Vd
SC C-16	限制驾驶员力 Limit pilot forces
SC D-03	应急出口布置—外部观察 Emergency exit arrangement-outside viewing
SC D-04	机组休息舱 (TC 证后) Crew rest compartments (Post TC)
SC D-06	客舱之间的楼梯的使用 Use of stairs between decks
SC D-07	客舱的火探测和保护 Fire detection and protection in passenger cabin
SC D-12	保安设计 Design for security
SC D-28	协调 671/672 Harmonised 671/672
SC D-33	延长应急滑梯 Extendable length escape slide
SC D-39	动态座椅的惯性锁定装置 (选装) Inertia Locking Device in Dynamic Seats (optional)
SC D-41	套装型座椅的安装 (选装) Installation of Suite Type Seating (optional)
SC D-42	C 型乘客出口 (选装) Type C Passenger Exits (optional)
SC D-45	通向上层客场楼梯间的餐车存放/提升系统 Trolley Stowage/ Lift Systems with Proximity to Upper Deck Staircase
SC D-47	膨胀式安全带的安装 Installation of Inflatable Seat Belts
SC D-52	安装结构固定安全气囊 (可选) Installation of structure mounted airbag (optional)
SC D-55	安装淋浴 (可选) Shower installation (optional)
SC F-01	JAR 25.1301 和 1309 符合性: 设计保证和安全评估程序 JAR 25.1301 and 1309 compliance: Design assurance and safety assessment process
SC F-02	滑梯/救生筏可携带性 Slide/ Raft portability
SC F-12	HIRF 保护 HIRF Protection
SC F-26	飞行记录器, 数据链记录

	Flight recorders, data link recording
SC F-52	锂离子电池安装
	Lithium – Ion battery installation

4.2 根据 JAR 21.16(a)(2)，由于产品的预期使用是非常规的而颁发专用条件
Special Conditions issued because the intended use of the product is unconventional (JAR 21.16(a)(2)):

SC D-20	无拖把拖拽
	Towbarless towing
SC D-31	高高度运行
	High altitude operation
SC G-06	单发调机（选装）
	Ferrying one engine unserviceable (optional)

4.3 根据 JAR 21.16(a)(31)，由于其他产品的使用经验表明具有不安全状况而颁发专用条件
Special Conditions issued because experience from other products has shown that unsafe conditions may develop (JAR 21.16(a)(3)):

SC D-13	隔热和隔音材料的防火
	Fire protection of thermal and acoustic insulation material
SC D-15	刹车和刹车系统
	Brakes and braking system – NPA 25D291
SC D-43	座椅材料的热释放和烟雾浓度
	Heat Release and Smoke Density to Seat Materials
SC D-46	便携式电子设备充电积载
	PED Charging Stowage
SC E-02	油箱安全
	Fuel tank safety
SC E-04	反推系统要求
	Thrust reverser system requirements
SC E-05	发动机持续不平衡
	Sustained engine imbalance
SC H-01	EWIS 的持续适航文件
	ICA on EWIS

5. 等效安全 / Equivalent Safety Findings (JAR 21.21(c)(2)):

ESF C-12	振动、颤振和气弹稳定性要求
	Vibration, buffet and aeroelastic stability requirements
ESF C-14	结构证明
	Proof of structure
ESF C-19	检查的俯仰机动载荷
	Checked Pitching manoeuvre loads
ESF C-20	发动机失效载荷
	Engine failure loads
ESF C-21	持续紊流载荷
	Continuous turbulence loads
ESF D-17	机头舱门
	Fuselage doors
ESF D-19	铸造系数
	Casting factors

ESF D-21	飞机客舱中允许的二氧化碳浓度和客舱臭氧浓度 Allowable carbon dioxide concentration in aeroplane cabins and cabin ozone concentration
ESF D-24	组件包关闭的运行 Packs off operation
ESF D-48	机腹整流罩隔音/隔热材料 Belly Fairing Thermal/acoustic Insulation Materials
ESF D-49	增强的下层客舱机组可燃性标准 Improved flammability standards for Lower Deck crew
ESF D-50	复合材料压力隔框隔音/隔热材料 Composite Pressure Bulkhead Thermal/acoustic Insulation Materials
ESFD-56	与飞机中心线夹角超过 18° 的前朝向座椅 Forward facing seat with more than 18° to a/c centerline
ESF E-06	雪跌落和吹雪 Falling and blowing snow
ESF E-09	油箱适坠性 Fuel tank crashworthiness
ESF E-10	燃油箱接近口盖 Fuel tank access covers
ESF E-11	RR Trent 涡轮过热探测 (仅适用于 A380-841/-842 型别) Rolls-Royce Trent turbine overheat detection (for A380-841/-842 models only)
ESF E-15	发动机燃油过滤器警告方式 (仅适用于 A380-841/-842 型别) Warning means for engine fuel filters (for A380-841/-842 models only)
ESF E-16	反推测试 Thrust reverser testing
ESF E-17	滑油温度指示 Oil temperature indication
ESF E-20	灭火剂浓度-符合 JAR 25.1195(c) (TC 证后-只有 A380-841/-842) Fire extinguishing agent concentration – compliance with JAR 25.1195(c) (Post TC – A380-841/-842 models only)
ESF F-11	空气系统 Pneumatic systems
ESF F-15	液压系统 Hydraulic systems
ESF F-23	着陆灯开关 Landing light switch
ESF F-29	新的经协调的 JAR 25.1329 New Harmonised JAR 25.1329
ESF F-38	释压阀和排气阀 Overpressure relief valves and outflow valves
ESF F-48	使用计算机模拟和相似性方法来验证高能转子包容性 Use of computer simulation and similarity approach for high energy rotor containment demonstration
ESF J-02	APU 安装要求 APU installation requirements
ESF K-06	定位器过偏警告 Localizer excessive deviation alerts
ESF K-07	限制风险 Limit Risk (NPA AWO 14)

6.1 燃油排放和排气排出物 / Fuel venting and emissions:

国际民航组织附件 16 第 2 版第 2 卷第 4 修正案。第 2 章第 II 部分和第 III 部分。
ICAO annex 16 second edition volume 2 amdt. 4 part II and Part III chapter 2.

6.2 噪声 / Noise:

国际民航噪声标准 (附件 16, 第 1 卷, 第 7 修正案, 第 4 章第 II 部分)。
ICAO Noise Standard (Annex 16, Volume 1, Part II, Chapter 4, amendment 7)

7. 自愿满足 / Elect to Comply

根据 2001 年 7 月 20 日的空客公司信函 AI/LE-A 828.0005/99 第 3 版, 空客公司选择自愿满足 2003 年 5 月 1 日颁发的 JAR 25 第 16 修正案的下述条款, 以及 NPA 25B-G-311、25C-312、25D-F-313、25C-D-F-314、25E-315 和 NPA 25G-321:

The following paragraphs of JAR 25 at amendment 16 issued May 1st, 2003 and which includes NPA 25B-G-311, 25C-312, 25D-F-313, 25C-D-F-314, 25E-315 (better plan for harmonisation) and NPA 25G-321 (Omnibus NPA for subpart G) that are elected to comply by Airbus per their letter AI/LE-A 828.0005/99 iss.3 dated 20 July 2001:

JAR 25.21(d)	JAR 25.791	JAR 25.954	JAR 25.1321	JAR 25.1521(d)
JAR 25.25	JAR 25.803	JAR 25.961	JAR 25.1325 title	JAR 25X1524
JAR 25.149(e)	JAR 25.807	JAR 25.967	JAR 25.1415	JAR 25.1527
JAR 25.251	JAR 25.812	JAR 25.975(a)(5)	JAR 25.1441	JAR 25.1545
JAR 25X261	JAR 25.815	JAR 25.981	JAR 25.1443	JAR 25.1547
JAR 25.337	JAR 25.853	JAR 25.993	JAR 25.1445(a)	JAR 25.1549
JAR 25.493	JAR 25.857	JAR 25.994	JAR 25.1447	JAR 25.1581
JAR 25.562(b)	JAR 25.863(b)(4)	JAR 25.997	JAR 25.1449	JAR 25.1583
JAR 25.605	JAR 25.904	JAR 25.1013	JAR 25.1450	JAR 25.1585
JAR 25.607	JAR 25.907	JAR 25.1015	JAR 25.1457	JAR 25.1587
JAR 25.701	JAR 25.933	JAR 25.1019	JAR 25.1513	
JAR 25.733	JAR 25.939	JAR 25.1145	JAR 25X1516	
JAR 25.777	JAR 25.951	JAR 25.1303	JAR 25.1517	
JAR 25.781	JAR 25.952	JAR 25.1305	JAR 25.1519	

Appendix D paragraph (b)
Appendix H subparagraph H25.3(e)
Appendix I

注: JAR 25 第 16 修正案中的 JAR 25.1517 被等效安全 ESF C-21 修订。

Note: JAR 25.1517, as in amendment 16 of JAR 25, is amended by Equivalent Safety Finding ESF C-21.

根据 2001 年 7 月 20 日的空客公司信函 AI/LE-A 828.0005/99 第 3 版, 空客公司选择自愿满足 2003 年 2 月 7 日 JAAC 采纳的、由 NPA AWO 8 和 10 修订的 JAR AWO 的下述要求:
The following paragraphs of JAR AWO as modified per NPA AWO 8 and 10, adopted by the JAAC on February 7th, 2003, that are elected to comply by Airbus per their letter AI/LE-A 828.0005/99 iss.3 dated 20 July 2001:

JAR AWO 第 3 分部, 第 B 节, 第 3 段; JAR AWO 第 3 分部, 第 C 节, 第 2 段; JAR AWO 第 3 分部, 第 D 节, 第 1 段; JAR AWO 第 4 分部, 第 2 段

Introduction to JAR AWO Subpart 3, section B, 3rd paragraph, Introduction to JAR AWO Subpart 3, section C, 2nd paragraph, Introduction to JAR AWO Subpart 3, section D, 1st paragraph, Introduction to JAR AWO Subpart 4, 2nd paragraph

JAR AWO 131(c)(2)	JAR AWO 313	JAR AWO 316(a)	JAR AWO 381
JAR AWO 304(b)	JAR AWO 314	JAR AWO 321(c)(4)	JAR AWO 481(a)
JAR AWO 305	JAR AWO 316 title	JAR AWO 321(d)(4)	

8. CAAC 认可审定的申请日期 / Application Date for CAAC Validation:

A380-841/-842: 20 January 2006

9. CAAC 认可审定日期 / CAAC Validation Date:

A380-841/-842: 15 December 2009

10. CAAC 认可审定基础 / CAAC Validation Basis:

CAAC 认可审定基础包括 / The CAAC validation basis consists of:

(1) EASA 确定的审定基础; 和 / Certification basis as established by EASA; and

(2) 如下附加技术条件 / Additional Technical Conditions as follows:

(2.1) 3 号燃油 / No. 3 Jet fuel

(2.2) 标记标牌的语言要求 / Language requirement of literal marking and placards

(2.3) 驾驶舱舱门和保安事项 / pilot compartment doors and the security considerations

(2.4) 快速存储记录器要求 / Quick Access Recorder (QAR) requirement

(2.5) 米制高度表要求 / Altimeter should display flight altitude with the metric system

(2.6) CCAR 25.811 应急出口的标记 / Emergency exit marking

(2.7) CCAR 25.857 货舱等级 / Cargo compartment classification

(2.8) CCAR 25.858 货舱或行李舱烟雾或火警探测系统 / Cargo or baggage compartment smoke or fire detection systems

III. 技术特性和运行限制 / Technical Characteristics and Operational Limitations

四台涡扇发动机的长程、双通道、运输类飞机
Four turbo-fan, long range, twin-aisle, large category airplane.

1. 装有罗罗发动机的 A380-841/-842 / A380-841/-842 Powered by RR Engines

1.1 型号设计定义 / Type Design Definition:

型号设计 / Type Design:

A380-841: 00L000H0841/C0S, Issue 3, October 2007

A380-842: 00L000H0842/C0S, Issue 1, December 2006

1.2 描述 / Description

4 台涡扇发动机、远程、双通道大型运输类飞机。

Four turbo-fan, long range, twin-aisle, large category airplane.

1.3 发动机 / Engines:

A380-841: Four (4) RB211 Trent 970-84 or RB211 Trent 970B-84 turbofan engines
 A380-842: Four (4) RB211 Trent 972-84 or RB211 Trent 972B-84 turbofan engines

发动机限制 / Engine Limits:

发动机限制 数据单 EASA E-012 CAAC VTC 0209E ENGINE LIMITS DATA SHEET EASA E-012 CAAC VTC 0209E	A380-841 RB211 Trent 970B-84	A380-842 RB211 Trent 972B-84
海平面静推力 - 起飞(5mn)* (保持额定推力的 极限环境温度为 30 度) Static thrust at sea level: - Take-off (5mn)* (flat rated 30°C)	348.31 kN	356.81 kN
经批准的滑油 / Approved Oil	参见发动机使用指南中关于 Trent 900 系列的经批准的滑油规范 的信息 Refer to the Engine Operating Instructions for information on approved oil specifications for the Trent 900	

* 根据 EASA TCDS 第 VI-1 段, 仅在发动机失效状态下 (在起飞或者复飞过程中) 允许使用 10 分钟起飞推力。
 10 minutes at take-off thrust allowed only in case of engine failure (at take-off or during go-around) in accordance with EASA TCDS paragraph VI-1.

其他发动机限制: 参见相关的发动机型号合格证数据单。
 Other engine limitations: See the relevant Engine Type Certificate Data Sheet.

1.4 燃油 / Fuel

名称 NOMENCLATURE	规范 / SPECIFICATION				
	法国 FRANCE	美国 U.S.A.	英国 U.K.	俄罗斯 RUSSIA	中国 CHINA
KEROSENE	DCSEA 134/B (JET A1) Kerosene	ASTM D-1655 (Jet A), (Jet A1)	DEF STAN 91-91 AVTUR	RJFS0 GOST 10227 - 86, (RT) (TS-1)	PRC MPIS GB 6537-2006 (No3 Jet Fuel)
		USA Air Force MIL-DTL- 83133 (JP8)	DEF STAN 91-87 AVTUR FSII	FAoTR&M GOST R 52050-2006 (JET A1)	
	DCSEA 144(JP 5)	US Navy MIL-DTL-	DEF STAN 91-86		

	Kerosene	5624(JP5)	AVTUR FSII		
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添加剂 参见罗罗安装手册“RB211 Trent 900 的专用运行要求”。上述的燃油和添加剂也适用于 APU

Additives See Rolls Royce “RB211 Specific Operating Instructions for Trent 900”, installation manual. The above-mentioned fuels and additives are also suitable for the APU.

1.5 速度限制 / Limit Speeds

参见经批准的飞机飞行手册。 / Refer to approved Airplane Flight Manual.

1.6 重心范围 / Centre of Gravity Range

参见经批准的飞机飞行手册。 / Refer to approved Airplane Flight Manual.

1.7 最大审定重量 / Maximum Certified Weights

VARIANT (Modification Number)	000 Basic	001 (64636)	002 (64605)	003 (66611)	004 (69436)	005 (69879)	006 (73786)	007 (71127)	008 (73787)
MTW (T)	562	512	571	512	562	562	575	492	577
MTOW (T)	560	510	569	510	560	560	573	490	575
MLW (T)	386	394	391	395	391	386	393	395	394
MZFW (T)	361	372	366	373	366	366	368	373	369

1.8 注释 / Notes

2. 保留 / Reserved

3. 适用于 A380-800 全部系列的数据 / Data Pertinent to all A380-800 Series

3.1 设备 / Equipment

必须安装适用规章要求的设备。

The equipment required by the applicable requirements shall be installed.

客舱设施、设备和布局必须符合下述规范：

Cabin furnishings, equipment and arrangement shall conform to the following specification:

- 00L252C0028/C01 对于客舱座椅 / for cabin seats,
- 00L252C0027/C01 对于厨房 / for galley,
- 00L252C0032/C01 对于客舱乘务员座椅 / for cabin attendant seats.

3.2 辅助动力装置 / Auxiliary Power Unit

一台加普惠 PW980A。

One Pratt & Whitney Canada PW980A

滑油：参见经批准的手册。

Oils: Refer to applicable approved Manuals

3.3 螺旋桨 / Propellers

无 / None

3.4 燃油量 / Fuel Quantity

油箱 / Tanks		可用燃油 升 (公斤) Usable Fuel Litres (Kg)	不可用燃油 升 (公斤) Unusable Fuel Litres (Kg)
机翼 Wing	左外侧 / Outer Left	10 340 (8 272)	38 (30)
	1 号 / Feed 1	27 632 (22 106)	82 (66)
	左中 / Mid Left	36 461 (29 169)	50 (40)
	2 号 / Feed 2	46 142 (36 914)	70 (56)
	3 号 / Feed 3	29 349 (23 479)	88 (70)
	右内侧 / Inner Right	46 142 (36 914)	70 (56)
	右中 / Mid Right	36 461 (29 169)	50 (40)
	4 号 / Feed 4	27 632 (22 106)	82 (66)
	右外侧 / Outer Right	10 340 (8 272)	38 (30)
配平 Trim		23 698 (18 958)	49 (39)
系统 Systems		793 (634)	382 (305)
总量 / Total		324339 (259471)	1086 (869)

3.5 飞行包线 / Flight Envelope

参见经批准的飞机飞行手册。 / Refer to approved Airplane Flight Manual.

3.6 其他限制 / Other Limitations

参见经批准的飞机飞行手册。 / Refer to approved Airplane Flight Manual.

3.7 全天气能力 / All Weather Capabilities

飞机具备 3 类精密进近和自动着陆能力。
The aircraft is qualified to Cat 3 precision approach and autoland.

3.8 最小飞行机组 / Minimum Flight Crew:

2 人 / Two (2): 驾驶和副驾驶 / Pilot and Co-pilot

3.9 最大座位数 / Maximum Seating Capacity

应急撤离批准的最大乘客数为 853 人 / The maximum number of passengers approved for emergency evacuation is: 853

- 上层客舱 / Upper deck: 315 乘客 / pax
- 主客舱 / Main deck: 538 乘客 / pax

交付时每架飞机批准的最大乘客数参见客舱布局图纸。
See interior layout drawing for the maximum passenger capacities approved for each aeroplane when delivered.

3.10 货舱装载 / Cargo Compartment Loading

货舱 / Cargo compartment	最大载荷 (公斤) / Maximum load (kg)
前 / Forward	28577 Kg or 63000 lb
后 / Aft	20310 Kg or 44775 lb
尾部 / Rear (bulk)	2515 Kg or 5540lb

对于每一位置允许的的定位和装载（根据集装箱、货物托盘和相应重量），参见下述文件的重量和平衡第 1.10 章：

For the positions and the loading conditions authorized in each position (references of containers, pallets and associated weights), see Weight and Balance Manual Chapter 1.10 ref.:

- 00L080H0001/C0S

3.11 机轮和轮胎 / Wheels and Tyres

轮胎混装：允许的组合参见服务通告 A380-32-8021（起落架—轮胎—总体程序）

Tyres mixability: See Service Bulletin A380-32-8021 (Landing Gear – Tires –General Prodedures) for allowable combinations.

3.12 电源中心构型数据文件工具 / Electrical Power Center Configuration Data File Tool

开发和经合格鉴定的航空公司构型工具(ACTC)用于航空公司管理次级电源分配盒(SPDB)的构型数据文件。该 ACTS 工具必须根据 SIL “电气系统构型数据文件更新” (文件号 SIL 24-085)使用。An Airline Configuration Tool (ACTS) has been developed and qualified to allow airlines to manage the Configuration Data File of Secondary Power Distribution Boxes (SPDB). This ACTS tool shall be used in accordance with the SIL “Guidance on Electrical system Configuration Data File update” reference “SIL 24-085”.

ACTS 工具的适用版本为第 2 版(CSCI 51220010-7)。

Applicable version of the ACTS tool is version 2 (CSCI 51220010-7)

IV. 使用和运行要求 / Operating and Service Instructions

1. 经批准的飞机飞行手册 / Approved Aircraft Flight Manual

经批准的飞机飞行手册 / Approved Aircraft Flight Manual: STL 38000

2. 持续适航文件和适航限制

Instructions for Continued Airworthiness and Airworthiness Limitations

- 在 A380 适航限制章节第 1 部分第 08 版（EASA 批准的文件 00L050HLS01/C01^[1]）中提供适用于安全寿命适航限制项目的限制。
Limitations applicable to **Safe Life Airworthiness Limitation Items** are provided in the A380 Airworthiness Limitations Section Part 1, Revision 08 (Document 00L050HLS01/C01, approved by EASA ^[1]).
- 在 A380 适航限制章节第 2 部分第 04 版（EASA 批准的文件 00L050H0005/C01^[1]）中提供适用于损伤容限适航限制项目的限制。
Limitations applicable to **Damage-Tolerant Airworthiness Limitation Items** are provided in the A380 Airworthiness Limitations Section Part 2, Revision 04 (Document 00L050H0005/C01, approved by EASA ^[1]).

- 在 A380 适航限制章节第 3 部分第 03 版 (EASA 批准的文件 00L050HLS03/C01^[1]) 中提供适用于审定维修要求的限制。
Limitations applicable to **Certification Maintenance Requirements** are provided in the A380 Airworthiness Limitations Section Part 3, Revision 03 (Document 00L050HLS03/C01, approved by EASA^[1]).
- 在 A380 适航限制章节第 4 部分第 09 版 (EASA 批准的文件 00L050HLS04/C01^[1]) 中提供适用于老龄系统维修的限制。
Limitations applicable to **Ageing System Maintenance** are provided in the A380 Airworthiness Limitations Section Part 4, Revision 09 (Document 00L050HLS04/C01, approved by EASA^[1]).
- 在 A380 适航限制章节第 5 部分第 04 版 (EASA 批准的文件 00L050H0009/C01^[1]) 中提供适用于燃油适航限制的限制。
Limitations applicable to **Fuel Airworthiness Limitations** are provided in the A380 Airworthiness Limitations Section Part 5, Revision 04 (Document 00L050H0009/C01, approved by EASA^[1]).
- 在 A380 适航限制章节第 6 部分第 04 版 (EASA 批准的文件 00L050HLS06/C01^[1]) 中提供适用于飞机信息系统保安的限制。
Limitations applicable to **Aircraft Information System Security** are provided in the A380 Airworthiness Limitations Section Part 6, Revision 04 (Document 00L050HLS06/C01, approved by EASA^[1]).
- 维修评审委员会报告 / **Maintenance Review Board Report** 00L050H0001/C01

注释[1]: 包括最新经批准的后续版本和 ALS 变化
Note [1]: Including last approved revision and ALS variations

V. 注释 / Notes

无 / None

- 完 / END -